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### Authors

San To Chan  
Ault, Jesse T  
Haward, Simon J  
[et al.](#)

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# Coupling of vortex breakdown and stability in a swirling flow

San To Chan,<sup>1,\*</sup> Jesse T. Ault,<sup>2,\*</sup> Simon J. Haward,<sup>1</sup> E. Meiburg,<sup>3</sup> and Amy Q. Shen<sup>1,†</sup>

<sup>1</sup>*Okinawa Institute of Science and Technology Graduate University, Onna, Okinawa 904-0495, Japan*

<sup>2</sup>*Biomedical Sciences, Engineering, and Computing Group, Oak Ridge National Laboratory, Oak Ridge, TN 37831, USA*

<sup>3</sup>*Department of Mechanical Engineering, University of California, Santa Barbara, Santa Barbara, California 93106, USA*

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Swirling flows are ubiquitous over a large range of length scales and applications including micron-scale microfluidic devices up to geophysical flows such as tornadoes. As the viscous dissipation, shear and centrifugal stresses interact, such flows can often exhibit unexpected fluid dynamics. Here, we use microfluidic experiments and numerical simulations to study the flow in a vortex T-mixer: a T-shaped channel with staggered, offset inlets. The vortex T-mixer flow is characterized by a single dominant vortex, the stability of which is closely coupled to the appearance of vortex breakdown. Specifically, at a flow Reynolds number of  $Re \approx 90$ , a first vortex breakdown region appears in the steady state solution, rendering the vortex pulsatively unstable. A second vortex breakdown region appears at  $Re \approx 120$ , which restabilizes the vortex. Finally, a third vortex breakdown region appears at  $Re \approx 180$ , which renders the vortex helically unstable. Thus, a counter-intuitive flow regime exists for the vortex T-mixer in which increasing the flow Reynolds number has a stabilizing effect on the steady state flow. The pulsatively unstable vortex evolves into a periodically pulsating state with a Strouhal number of  $St \approx 0.5$ , and the helically unstable vortex evolves into a helically oscillating state with  $St \approx 1.75$ . These transitions can be explained within the framework of linear hydrodynamic stability. In addition, the vortex T-mixer flow exhibits multi-stability; multiple flow states are stable over various ranges of  $Re$ , including a narrow range of tri-stability for  $160 \leq Re \leq 170$  in which the steady state, the pulsatile oscillation, and the helical oscillation are all stable. This study provides experimental and numerical evidence of the close coupling between vortex breakdown and flow stability, including the restabilization of the flow with increasing Reynolds number due to the appearance of a vortex breakdown region, which will provide new insights into how vortex breakdown can affect the stability of a swirling flow.

## I. INTRODUCTION

Swirling flows are flows which rotate about an axis; a typical example is tornado. They exhibit a diverse range of fluid dynamical behaviors as the viscous dissipation, shear and centrifugal stresses interact. For instance, strong enough centrifugal stress can trigger a phenomenon called vortex breakdown, which corresponds to the sudden structural change of vortex core of a swirling flow; for example, the sudden divergence of stream surfaces [1,2].

Two predominant types of vortex breakdown exist: the bubble-type and the spiral-type [3]. Bubble-type breakdown is characterized by a nominally axisymmetric flow recirculation zone that is bounded by upstream and downstream internal stagnation points. If a tracer filament is injected on the vortex axis, it will become trapped in the recirculation zone and appear like a bubble [4]. Spiral-type breakdown is instead characterized by a single internal stagnation point followed by a helical motion of the vortex core downstream. In this case, a tracer filament will kink at the location of the stagnation point and flutter helically downstream [5]. Spiral-type breakdown can be further characterized into the single and double forms. The single form has only one fluttering helix; in contrast, the double form has two. Direct numerical simulations [6–8] have revealed that spiral-type breakdown is caused by a large pocket of absolute instability that is located around the bubble-type breakdown region. Thus, the bubble-type is the more basic form of vortex breakdown.

Both types of vortex breakdown are of vital importance to the understanding of natural phenomena and to the performance of engineering systems. For example, vortex breakdown observed in model tornadoes plays a crucial role in inferring the internal structure and evolution of real tornadoes [9,10]. In combustion systems, recirculation zones due to bubble-type breakdown can help fuels burn more effectively and create a compact, stable flame [11–13]. In aerodynamics systems, such as flow over a delta wing, depending on the angle of attack breakdown can occur in the leading-edge vortices, which can drastically alter stability of the aircraft [14,15]. Recently, vortex breakdown has seen applications in microfluidics due to its ability to manipulate micro-particles in a branching T-junction flow [16–21], in which one inflow splits into two opposite outflows.

Here, we combine microfluidic experiments and numerical simulations to study the mixing flow in a T-shaped channel with staggered, offset inlets, which we will refer to as a *vortex T-mixer* (Fig. 1). This flow geometry was chosen for two main reasons. First, unlike the branching T-junction flow which has four Dean vortices [16–21], the vortex T-mixer flow has only one single dominant vortex, which greatly simplifies the interpretation of the numerical and experimental results. Second, it is easy to perform measurements for the vortex T-mixer flow. For instance, the flow field can be quantitatively measured directly at the channel cross-section, allowing direct comparison between the experimental and numerical results.

We will show that the vortex T-mixer flow exhibits stability characteristics that are tightly coupled to the appearance and evolution of vortex breakdown regions. In particular, at a flow Reynolds number (which quantifies the relative inertial and viscous effects of a flow) of  $Re \approx 90$ , a single

\* Authors contributed equally to this work.

† amy.shen@oist.jp

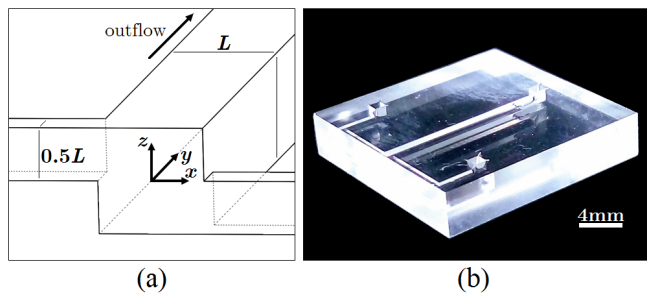


FIG. 1. Microfluidic T-mixer with staggered, offset inlets (vortex T-mixer). (a) Coordinate system and schematic of the vortex T-mixer. (b) Experimental devices are fabricated in glass using a LightFab 3D-printer. Both inflows enter the junction tangentially, resulting in the development of a swirling flow characterized by a single, dominant vortex structure.

bubble-type vortex breakdown region appears in the steady state flow. The appearance of this region destabilizes the flow to pulsative perturbations and triggers the transition of the steady state flow to a stable periodically pulsating state with a Strouhal number (which is the dimensionless frequency of an oscillation) of  $St \approx 0.5$ . At  $Re \approx 120$ , a second vortex breakdown region appears that restabilizes the steady state flow. Thus, the vortex T-mixer flow exhibits counter-intuitive stability characteristics in this Reynolds number range in which increasing the Reynolds number actually has a stabilizing effect on the flow. Finally, a third vortex breakdown region appears at  $Re \approx 180$  that renders the flow unstable to helical perturbations, triggering the steady state to undergo spiral-type breakdown and resulting in a stable helical oscillating state with  $St \approx 1.75$ . In addition, the vortex T-mixer flow shows a strong history dependence. That is, depending on how  $Re$  is varied, multiple flow states can be stable over different  $Re$  ranges, including a narrow range of  $160 \leq Re \leq 170$  in which the steady, pulsatile and helical states are all stable.

In the following sections we present experimental and numerical evidence of this close coupling between the formation of vortex breakdown in the vortex T-mixer and the stability characteristics of the flow, which will point to new research directions for the study of vortex breakdown. In §II we describe the experimental and numerical methodologies used in this analysis, in §III we present results and discussion, and in §IV we present conclusions and ideas for future work.

## II. METHODOLOGY

In this section, we describe the experimental methodology used to fabricate our devices and perform the experiments, we introduce the notations that will be used throughout, and we describe the numerical approach used to simulate the T-mixer flow.

### A. Microfluidic device

All experiments were performed in a vortex T-mixer as shown in Fig. 1(a). Such channel geometry contains a  $Z_2$  symmetry [22], as it is invariant to a rotation of angle  $\pi$

about the  $y$ -axis. The choice to offset the inlets in this geometry was originally proposed as an approach to enhance the mixing performance of a simple T-shaped microchannel [23,24]. In this vortex T-mixer, both inflows enter the outlet channel tangentially, resulting in a swirling flow characterized by one dominant vortex structure that decays downstream in the channel outlet.

The vortex T-mixer was fabricated in glass using a LightFab 3D-printer (LightFab GmbH, Germany), which utilizes Selective Laser-induced Etching (SLE) technology [25,26]. The LightFab 3D-printer employs ultrafast laser pulses to print channels inside a piece of fused silica. The channels are then etched with potassium hydroxide (KOH) in an  $80^\circ\text{C}$  ultrasonic bath at an approximate rate of  $100 \mu\text{m}/\text{h}$ . Etching of the channels is about 1000 times faster than that of the bulk glass. The final product is a transparent, rigid piece of glass with embedded channels that can endure both organic solvents and high pressure (see Fig. 1(b)). Four of the six outer surfaces of the etched device are opaque; however, optical access can be gained by adding a thin film of water onto those surfaces. The outlet channel has a length of 20 mm and a square cross-section with side lengths of  $L = 1 \pm 0.06$  mm. It is sufficiently long such that the flow is insensitive to the outlet conditions [27,28]. The inlet channels have square cross-sections with side lengths of  $0.5 \pm 0.01$  mm and lengths of 8 mm, which ensures that the flow is fully developed before reaching the junction.

### B. Notations

Here, we briefly summarize our notations. We define the flow Reynolds number as  $Re = UL/\nu$ , where  $U$  is the mean flow speed in the outlet channel,  $\nu$  is the kinematic viscosity of the fluid, and  $L$  is the side length of the channel outlet. The coordinates, flow velocities and time are non-dimensionalized by  $L$ ,  $U$  and  $L/U$ , respectively. Those dimensionless variables are denoted as  $\mathbf{x} = (x, y, z)$ ,  $\mathbf{u} = (u_x, u_y, u_z)$  and  $t$ . For flows that oscillate periodically, the Strouhal number  $St = fL/U$  is used to non-dimensionalize the oscillation frequency  $f$ . The term *flow state* refers to the set of  $Re$  for which the flow oscillation can be approximately characterized by a single  $St$ .

### C. Flow rate control

The flow was driven by three individually controlled neMESYS syringe pumps (Cetoni GmbH, Germany) equipped with 25 ml glass syringes (Hamilton Gastight, Reno, NV). The pumps operate at a minimum of  $5\times$  (typically  $50\times$ ) the specified pulsation-free flow rate in order to eliminate pump-induced vibrations. The microfluidic device and syringes were connected by rigid polyethylene tubing to minimize hydraulic compliance. As the flow state depends on how  $Re$  is varied, two different approaches were used to initialize a desired  $Re$  in an experiment. The first is to directly increase  $Re$  from zero to the desired value in approximately 0.1 s, which corresponds to a dimensionless rate of  $\mathcal{O}(10)$ . The second is to slowly vary  $Re$  at a rate of approximately  $1 \text{ s}^{-1}$ , which corresponds to a dimensionless rate of  $\mathcal{O}(0.01)$ , from a reference value  $Re_1$  to the desired

value  $Re_2$ . With these two different initialization strategies we can demonstrate the multi-stability in which multiple stable solutions exist for a given  $Re$ .

#### D. Fluorescent dye visualization

Fluorescent dye visualization was performed on the channel cross-section at  $y = 2$ . To achieve this, one of the T-mixer inflows was supplied with pure water, while the other was supplied with a 0.05 g/L sodium fluorescein solution, which was illuminated by a metal halide lamp with a 488 nm excitation filter. Videos were captured by an inverted microscope (Nikon ECLIPSE Ti) equipped with a spinning-disk confocal imaging system (DSD2, Andor Technology Ltd) and a 0.10 NA, 30 mm WD, and a 4 $\times$  objective (Nikon CFI Achrom 4X) with an exposure time of 20 ms.

#### E. Micro-particle image velocimetry ( $\mu$ PIV)

Time-resolved micro-particle image velocimetry ( $\mu$ PIV) [29] was performed on the channel cross-section at  $y = 2$ . Red fluorescent polystyrene particles of diameter 2  $\mu$ m and density 1.05 g/cm<sup>3</sup> were seeded in deionized water and illuminated by a 527 nm dual-pulsed Nd:YLF laser (Terra PIV, Continuum Inc., CA) with an average power of 60 W and a pulse duration  $\delta t < 10$  ns. Depending on the considered  $Re$ , the laser repetition ranged from 500 to 1630 Hz, and the time separation between two laser pulses ranged from 5 to 100  $\mu$ s. The  $\mu$ PIV images were captured by an inverted microscope (Nikon ECLIPSE Ti-S) equipped with a 1280  $\times$  800 pixels high-speed CMOS camera (Phantom Miro M310, Vision Research Inc., NJ) and a 0.15 NA, 23.5 mm WD, 5 $\times$  objective (Nikon CFI Plan Fluor 5X). The measurement depth [30] over which the fluorescent particles contribute to the determination of the flow field was approximately 0.1 $L$ . To accurately compute  $St$  for the flow oscillation, at least 400  $\mu$ PIV images were captured for each  $Re$ . Dye visualization videos were also captured by the same high speed camera at 4000 and 10000 Hz to ensure that the  $\mu$ PIV measurements satisfied the Nyquist-Shannon sampling criterion [31].

#### F. Numerical simulation

Simulations were performed using the open source computational fluid dynamics software OpenFOAM [32]. Steady state simulations were performed using the built-in solver `simpleFoam`, and transient simulations were performed using a modified version of the `icoFoam` solver. Fully-developed inlet velocity profiles were implemented using the `groovyBC` boundary condition utility provided by the `swak4Foam` package, **which corresponds to the experimental inflow condition**. The selected solvers are second-order accurate in both space and time. A simulation domain that corresponds to the microfluidic device was generated using the built-in `blockMesh` utility. All simulations were performed using the same mesh with inlet channels of length  $4L$  and an outlet channel of length  $15L$ . **The outlet channel was determined to be sufficiently long such that the fluid dynamics of the system is**

**insensitive to the outlet boundary condition [27,28], and the outflow is approximately fully developed up to the maximum  $Re$  used in this study, which correspond to the experimental outflow condition.**

The boundary conditions imposed on the fluid velocity include a fully developed inflow condition at each channel inlet, no-slip conditions on all channel walls, and a zero-normal-gradient condition at the channel outlet. This implicitly assumes that the flow is fully developed when it reaches the channel outlet. Boundary conditions on the fluid pressure were selected to be zero-normal-gradient at the channel inlets and at the channel walls, as well as a fixed uniform outlet pressure of  $p = 0$ . This inlet boundary condition on the pressure, while not analytically correct, is helpful for the stability of the solvers, and the errors introduced by it are confined to a narrow region within one or two cell lengths of the channel inlets. Additional descriptions of the simulation domain, mesh design and inflow condition, as well as the results of a numerical convergence study are presented in Appendix A.

#### G. Stability of steady state solutions

Steady state simulations were performed for  $Re$  ranging from 10 to 250 by increments of 10. Residual tolerances were set to  $10^{-6}$  for both the fluid pressure and velocity; relaxation factors of 0.3 and 0.7 were respectively chosen for the pressure and velocity. Using this setup, the steady state solutions up to  $Re = 220$  successfully converged in less than one hour while running on 64 cores on the Titan supercomputer. These simulations were initialized using initial conditions of  $\mathbf{u} = \mathbf{0}$  and  $p = 0$  everywhere. To test the flow stability, converged steady state solutions were imported as initial conditions into a transient solver. These simulations were carried on until velocity measurements at probed locations were confirmed to be steady or until they achieved stable periodic orbits. Subsequently, additional simulations were performed to determine the  $Re$  limits of transition between states using the steady state and the stable oscillating solutions as initial conditions with incrementally varied  $Re$  until a new periodic orbit was achieved or the flow transitioned back to steady.

### III. RESULTS AND DISCUSSIONS

#### A. Unstable-stable transition

To motivate a systematic analysis of the fluid dynamics in a vortex T-mixer, we first present an unexpected experimental observation that for the same geometry and with steady inlet conditions, the laminar steady state base flow, which is unstable for approximately  $80 < Re < 120$ , can be restabilized by increasing the Reynolds number. Experimentally, this effect was observed using dye visualizations as described in the previous section and initializing the flow rate on the syringe pumps directly to the target Reynolds number. For each  $Re$ , a dye visualization video with at least 100 frames was time-averaged to form a single image as shown for  $Re = 80, 100, 130, \text{ and } 180$  in Fig. 2(a).

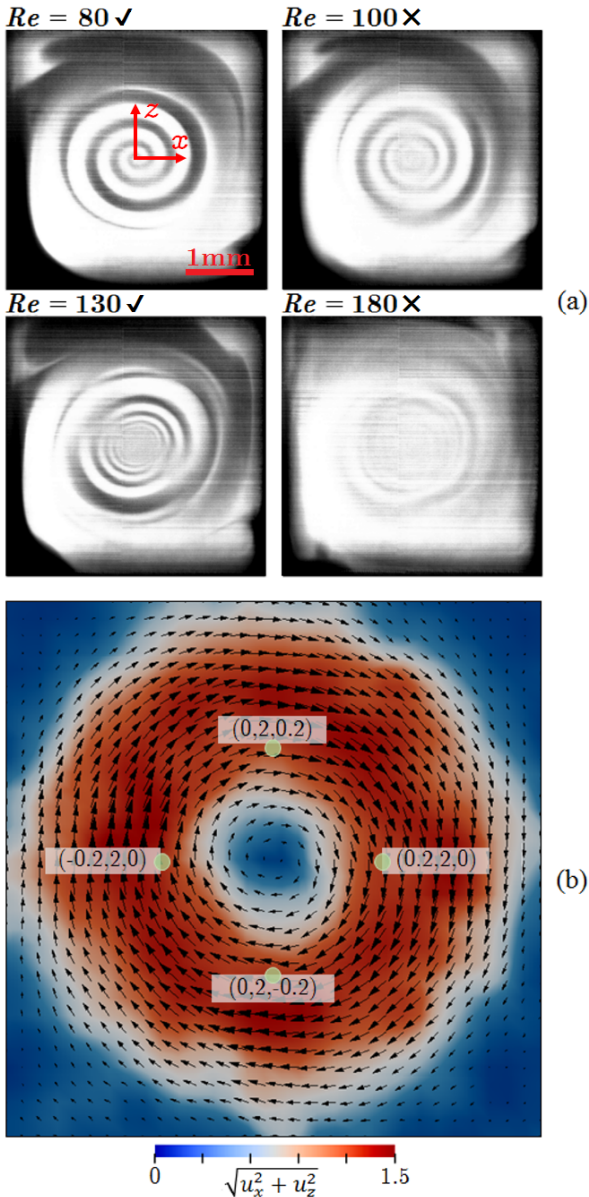


FIG. 2. Example experimental results on the channel cross-section  $y = 2$ . (a) Time-averaged fluorescent dye visualizations at different Reynolds numbers. A clear spiral between the two inflows of water and dye indicates a steady flow, as labeled by check marks, and a blurry spiral indicates an unsteady flow, as labeled by a cross. A corresponding video can be found in the supplemental information [33]. (b) A snapshot of the time-resolved micro-particle image velocimetry ( $\mu$ PIV) results at  $Re = 100$ .

For steady state flows, the time-averaged dye visualization should contain a clear interface between the two inflows of water and dye, whereas for unsteady flows this interface will be blurry due to the averaging over the video frames. In Fig. 2(a) we see that for  $Re = 80$ , a spiral with clear winding can be observed at the channel center, signifying a steady flow. **Handedness of such spiral is solely determined by orientations of the two inlet channels. Hence, it is independent to the initial condition applied.** Increasing the Reynolds number to  $Re = 100$ , the spiral blurs out, suggesting that the flow has transitioned to unsteady. Surprisingly, at  $Re = 130$  a clear spiral is once again observed,

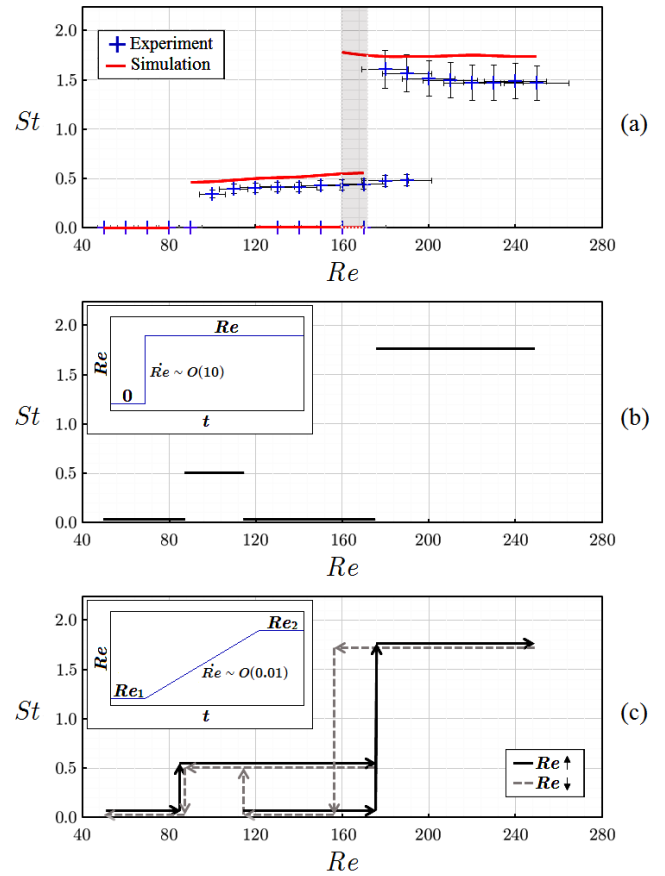


FIG. 3. St-Re flow state diagram. (a) Strouhal number  $St$  as a function of  $Re$ . Experimental and numerical results are respectively labeled as blue crosses and red lines. The vortex T-mixer flow can be characterized into three states, with  $St = 0$ ,  $St \approx 0.5$  and  $1.75$ . (b) A schematic showing how the flow state changes when  $Re$  is directly increased from zero to the desired value. (c) A schematic showing how the flow state changes when  $Re$  is varied from a reference value  $Re_1$  to a desired value  $Re_2$  linearly at a slow rate. Black solid and gray dashed arrows respectively indicate the increasing and decreasing of  $Re$ .

suggesting that the flow has regained its steadiness. Finally, at  $Re = 180$  the spiral almost completely disappears as the flow transitions back to unsteady once again. Thus, the vortex T-mixer flow demonstrates a counter-intuitive flow regime within which increasing the Reynolds number can restabilize the steady state base flow. A video corresponding to the results presented in Fig. 2(a) can be found in the supplemental information [33].

## B. St-Re flow state diagram

Motivated by the experimental observation of this unexpected restabilization, we performed a systematic experimental and numerical study of the different flow regimes, stability characteristics, and flow transitions observed in the vortex T-mixer flow. These results can be summarized in a St-Re state diagram as shown in Fig. 3. For each  $Re$ , the Strouhal number was obtained from a time series of  $u_z$  probed at the location  $\mathbf{x} = (0, 2, 0.2)$ . These results were obtained using both time-resolved  $\mu$ PIV [Fig. 2(b)] and transient simulations as described in the methodology section.

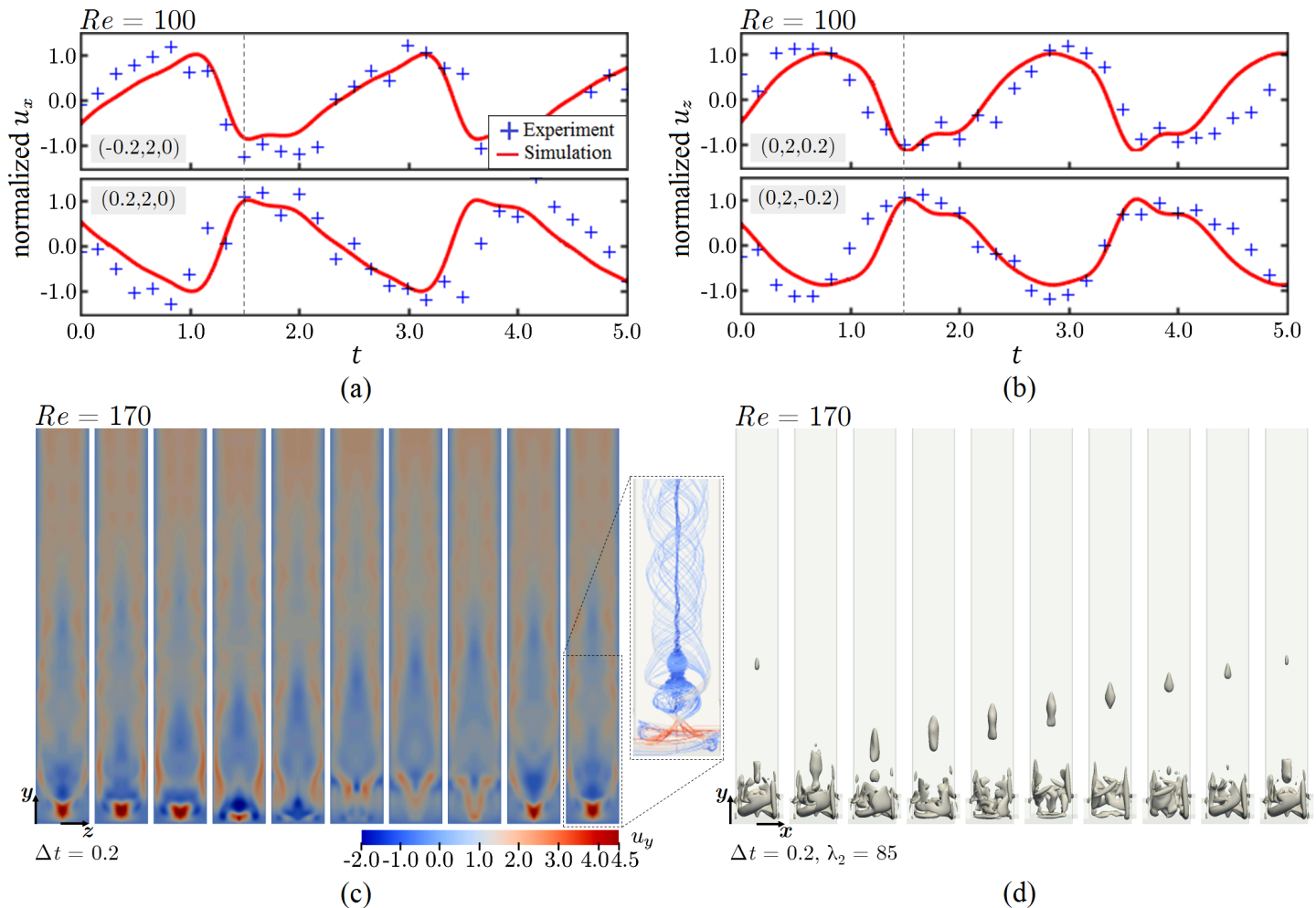


FIG. 4. The  $St \approx 0.5$  state. (a) Flow velocity  $u_x$  oscillating out-of-phase at locations  $\mathbf{x} = (-0.2, 2, 0)$  and  $(0.2, 2, 0)$ . (b) Flow velocity  $u_z$  oscillating out-of-phase at locations  $\mathbf{x} = (-0.2, 2, 0)$  and  $(0.2, 2, 0)$ . The four locations correspond to the left, right, top and bottom sides on the channel cross-section at  $y = 2$ . Blue crosses and red lines represent the experimental and numerical results, respectively. (a) and (b) are results with  $Re = 100$  and show that the flow is undergoing a non-axisymmetric pulsatile motion, which is analogous to the compressing and stretching of a hula hoop along one direction. (c) Axial velocity  $u_y$  on the channel mid-plane  $x = 0$ . Distributions of  $u_y$  are symmetric across the  $z = 0$  plane. The dashed box contains representative 3D streamlines that show a recirculation zone developing in the junction, which signifies bubble-type vortex breakdown. (d) A level set of the  $\lambda_2$  criterion [34,35] with  $\lambda_2 = 85$ . As can be seen, through each oscillation, a pocket of bubble-type vortex breakdown develops in the junction which is advected downstream where it ultimately decays. (c) and (d) are results with  $Re = 170$  and represent approximately one full cycle of the pulsatile motion with  $\Delta t = 0.2$  between frames. A corresponding video can be found in the supplemental information [33].

Fig. 3(a) shows the experimentally and numerically observed Strouhal numbers for the vortex T-mixer flow as a function of  $Re$ . Fig. 3(b) shows the flow state that is achieved when the flow rate is directly increased from zero to the target Reynolds number, while Fig. 3(c) shows the flow state that is achieved when the Reynolds number is gradually varied at a slow rate.

The  $St$ - $Re$  state diagram shown in Fig. 3 illustrates three distinguishing features of the vortex T-mixer flow. First, the vortex T-mixer flow can be characterized into three states, each of which has a distinct  $St$ , as shown by the three nominally horizontal red lines obtained by simulation in Fig. 3(a). The steady state solution with  $St = 0$  is seen to be stable for  $0 \leq Re \leq 80$  and  $120 \leq Re \leq 170$ , the  $St \approx 0.5$  state is stable for  $90 \leq Re \leq 170$ , and the  $St \approx 1.75$  state is stable for  $Re \geq 160$ . Second, the vortex T-mixer flow exhibits multi-stability, that is, within certain  $Re$  ranges multiple flow states are stable, including a region

of tri-stability. This regime appears in the Reynolds number range  $160 \leq Re \leq 170$ , where the steady  $St = 0$  state, the  $St \approx 0.5$  state, and the  $St \approx 1.75$  state are all stable. This is illustrated by the gray area in Fig. 3(a).

Finally, the  $St$ - $Re$  state diagram illustrates the history-dependence, i.e. hysteresis, of the vortex T-mixer flow as the Reynolds number is varied. When the flow rate is initialized directly to achieve the desired Reynolds number, a single flow state exists for each  $Re$  as shown in Fig. 3(b). Similarly, if the numerically obtained steady state solutions are used as initial conditions in a transient solver, the flow will transition to the  $St \approx 0.5$  state for  $90 \leq Re \leq 110$  and will remain in the steady state for  $120 \leq Re \leq 170$ . However, if  $Re$  is increased or decreased slowly from a reference value  $Re_1$  to the desired value  $Re_2$  as shown in Fig. 3(c), the flow states can persist for larger ranges of  $Re$ , giving rise to the multi-stability. For example, slowly increasing  $Re$  for the  $St \approx 0.5$  state can preserve this state up to  $Re \approx 170$ .

In the next sections, we will take an in-depth look at the three different identified flow states and elucidate the connection between the appearance of vortex breakdown regions and the dynamical shifts in stability of the flow.

### C. The $St \approx 0.5$ state

First, we consider dynamics of the  $St \approx 0.5$  state. Representative experimental and numerical results characterizing this flow state are shown in Fig. 4 for (a,b)  $Re = 100$  and (c,d)  $Re = 170$ . Time series data of  $u_x$  at  $\mathbf{x} = (-0.2, 2, 0)$  and  $(0.2, 2, 0)$ , and  $u_z$  at  $\mathbf{x} = (0, 2, 0.2)$  and  $(0, 2, -0.2)$  are presented in Fig. 4(a,b). These locations correspond to the left, right, top and bottom sides of the channel cross-section at  $y = 2$ . As can be seen,  $u_x$  oscillates out-of-phase on the left and right sides of the channel [Fig. 4(a)], and  $u_z$  oscillates out-of-phase on the top and bottom sides of the channel [Fig. 4(b)]. The right/left oscillations are seen to be out-of-phase with the top/bottom oscillations. Essentially, the flow is undergoing a non-axisymmetric, pulsatile motion analogous to the compressing and stretching of a hula hoop along one direction, **which preserves the  $Z_2$  symmetry of the vortex T-mixer**. As can be seen, there is close agreement between the numerical simulations and the experimental  $\mu$ PIV results.

In order to enable direct comparison between the different flow states, we also present detailed numerical results for the case of  $Re = 170$  at which the flow exhibits tri-stability. This will allow us to compare the steady state flow, the pulsating  $St \approx 0.5$  flow, and the helically oscillating  $St \approx 1.75$  flow under identical inflow conditions. Distributions of the axial velocity  $u_y$  on the channel mid-plane  $x = 0$  are shown in Fig. 4(c) as a function of time. Here, the flow fields are symmetric about the  $z = 0$  plane. In addition, bubble-type vortex breakdown is evidenced by a recirculation zone as shown by the 3D streamline visualization on the right side of Fig. 4(c). Fig. 4(d) shows a level set of the  $\lambda_2$  criterion, which is an algorithm for identifying vortices from three-dimensional velocity fields [34,35]. Here,  $\lambda_2 = 85$ . The results in Fig. 4(c,d) both correspond to approximately one full period of the stable oscillation. In each oscillation a single pocket of recirculating flow (i.e. a vortex breakdown bubble) is generated in the junction and then advected downstream, where it ultimately decays due to viscous dissipation.

### D. The $St \approx 1.75$ state

Next, we perform a similar characterization of flow for the  $St \approx 1.75$  state and contrast it with the pulsating state. Representative experimental and numerical results characterizing this flow state are shown in Fig. 5 for (a,b)  $Re = 180$  and (c,d)  $Re = 170$ . Once again, we consider  $u_x$  at  $\mathbf{x} = (-0.2, 2, 0)$  and  $(0.2, 2, 0)$ , and  $u_z$  at  $\mathbf{x} = (0, 2, 0.2)$  and  $(0, 2, -0.2)$  and plot the time series data in Fig. 5(a,b). In contrast to the pulsating  $St \approx 0.5$  state, here  $u_x$  oscillates in-phase on the left and right sides of the channel [Fig. 5(a)], as does  $u_z$  on the top and bottom sides of the channel [Fig. 5(b)]. Furthermore, the right/left oscillations here have a phase difference of  $\pi/2$  with the top/bottom os-

cillations. Thus, the flow is undergoing a periodic oscillating helical motion in the channel outlet, **which breaks the  $Z_2$  symmetry of the vortex T-mixer flow**.

As with the pulsating state, the experimental and numerical results show good agreement. To facilitate direct comparison between the two oscillating states, we again include detailed numerical results for  $Re = 170$  that falls within the regime of tri-stability. Fig. 5(c) shows the evolution of  $u_y$  on the  $x = 0$  channel center plane throughout a single complete oscillation of the flow. Unlike the  $St \approx 0.5$  state, here the oscillation clearly arises from a symmetry-breaking instability, and **the two-dimensional flow fields are no longer symmetric about the  $z = 0$  plane**. Within the junction, the vortex core is nominally axisymmetric, but this clearly breaks down and the vortex oscillates helically downstream, as shown by the streamline visualization in Fig. 5(c), signifying the onset of spiral-type vortex breakdown [3]. Finally, Fig. 5(d) shows a level set of the  $\lambda_2$  criterion again with  $\lambda_2 = 85$ . These results clearly distinguish the dynamics of the vortex breakdown regions between the two oscillating states. Whereas with the pulsating  $St \approx 0.5$  state the recirculation zone was produced in the junction and advected downstream along the channel outlet centerline, here several small pockets of recirculation appear to oscillate around the channel outlet centerline within a narrow range approximately  $1.5L$  to  $2.5L$  from the channel bottom. Furthermore, the breakdown does not appear to have any rotational symmetries, suggesting that the spiral-type vortex breakdown seen here is of the single form.

### E. The steady $St = 0$ state

As we have shown, the steady state flow, the pulsating  $St \approx 0.5$  flow, and the helically oscillating  $St \approx 1.75$  states are all stable solutions within the range of tri-stability given by  $160 \leq Re \leq 170$ . This suggests that the  $St \approx 0.5$  and  $1.75$  states arise from the steady state base flow through different instability mechanisms. In order to elucidate these instability mechanisms, we consider the evolution of the steady state base flow as a function of Reynolds number, and we especially emphasize the qualitative dynamical shifts that occur which are characterized by the appearance of vortex breakdown regions.

A summary of this systematic investigation of the steady state base flow showing the outlet centerline axial velocity component, streamline representations of the vortex breakdown regions, and the stability classification for a range of Reynolds numbers is presented in Fig. 6. Specifically, Fig. 6(a) shows how the dimensionless axial velocity component  $u_y$  varies along the outlet channel centerline for  $30 \leq Re \leq 190$ . Here,  $u_y$  is plotted against  $yRe$  for visualization purposes; doing so nicely collapses the data by effectively aligning the peaks of  $u_y$ . Vortex breakdown is indicated by the regions in which  $u_y < 0$  and is represented by the solid gray box. As can be seen, starting from  $Re = 30$ , increasing  $Re$  results in a local minimum of  $u_y$  that first goes negative around  $Re = 90$ , representing the onset of vortex breakdown. Varying the Reynolds number also results in a qualitative dynamical shift in the number of internal stagnation points. For example, for  $Re \leq 80$ , there are no internal stagnation points. However, at  $Re = 90$ ,

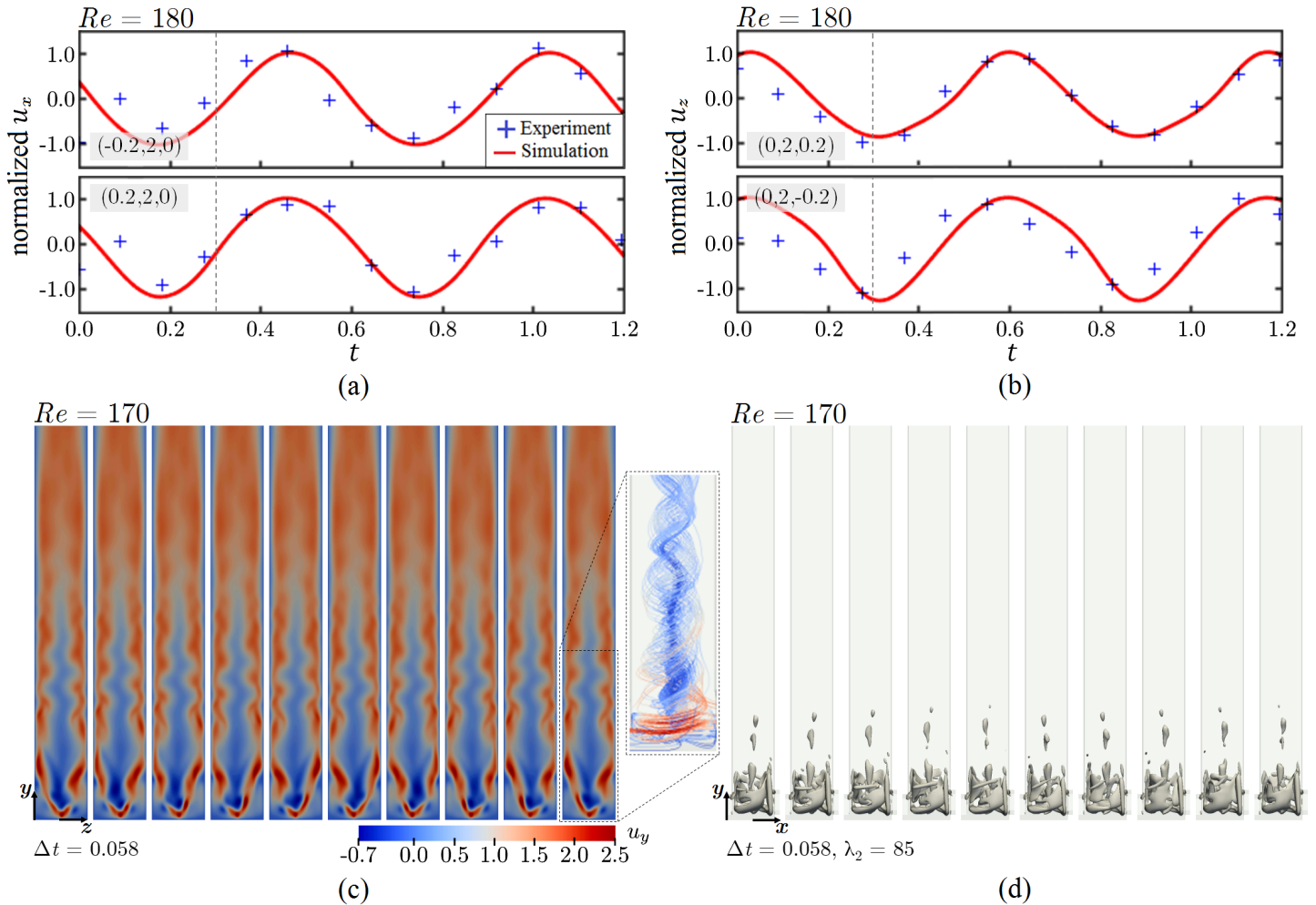


FIG. 5. The  $St \approx 1.75$  state. (a) Flow velocity  $u_x$  oscillating in-phase at locations  $\mathbf{x} = (-0.2, 2, 0)$  and  $(0.2, 2, 0)$ . (b) Flow velocity  $u_z$  oscillating in-phase at locations  $\mathbf{x} = (-0.2, 2, 0)$  and  $(0.2, 2, 0)$ . The four locations correspond to the left, right, top and bottom sides on the channel cross-section at  $y = 2$ . Blue crosses and red lines represent the experimental and numerical results, respectively. (a) and (b) are results with  $Re = 180$  and show that the left/right oscillations have a phase difference of  $\pi/2$  with the top/bottom oscillations, signifying that the flow is undergoing a periodic helical oscillation. (c) Distributions of  $u_y$  are asymmetric over the  $z = 0$  plane. The dashed box contains representative 3D streamlines that show that the vortex core is nominally axisymmetric in the junction but undergoes non-axisymmetric helical motion downstream, signifying the onset of spiral-type vortex breakdown. (d) A level set of the  $\lambda_2$  criterion [34,35] with  $\lambda_2 = 85$ . As can be seen, within the junction a single pocket of vortex, corresponding to the nominally axisymmetric vortex core, rotates around its own axis at a fixed position in the junction. Downstream, several small pockets of vortex helically oscillate about the channel centerline. (c) and (d) are results with  $Re = 170$  and represent approximately one full cycle of the helical oscillating motion with  $\Delta t = 0.058$  between frames. A corresponding video can be found in the supplemental information [33].

there are two internal stagnation points, also signifying the onset of bubble-type vortex breakdown.

Corresponding to the results in Fig. 6(a), 3D streamline visualizations of vortex breakdown regions in the outlet channel are shown as a function of  $Re$  in Fig. 6(b). They appear similar to those observed in a cylinder with a rotating end wall [36–38], which is a classical system for studying bubble-type vortex breakdown. At around  $Re = 90$ , a first bubble-type vortex breakdown region appears. Around  $Re = 120$ , a second region appears downstream of the first. As the Reynolds number increases further, these regions merge at around  $Re = 130$ . Finally, a third vortex breakdown region appears around  $Re = 180$ , enlarging and slenderizing the aforementioned second breakdown region. This region also merges into the existing breakdown structure at a slightly higher Reynolds number.

## F. Stability of the steady state solutions

Finally, we consider the stability and transient evolutions of the steady state base flow solutions, we document the instability mechanisms that trigger transition, and we emphasize the close connection between the stability and vortex breakdown characteristics for the vortex T-mixer. First, we test the stability of each of the steady state solutions as described in the methodology section, the results of which are shown in Fig. 6. As can be seen, every qualitative dynamical shift in the stability of the base state corresponds to a qualitative shift in the vortex breakdown characteristics of the flow. The base state is stable up until around  $Re = 90$ , where the first vortex breakdown region appears, triggering a pulsative instability in the vortex that causes a transition to the pulsating  $St \approx 0.5$  state. At around  $Re = 120$ , the



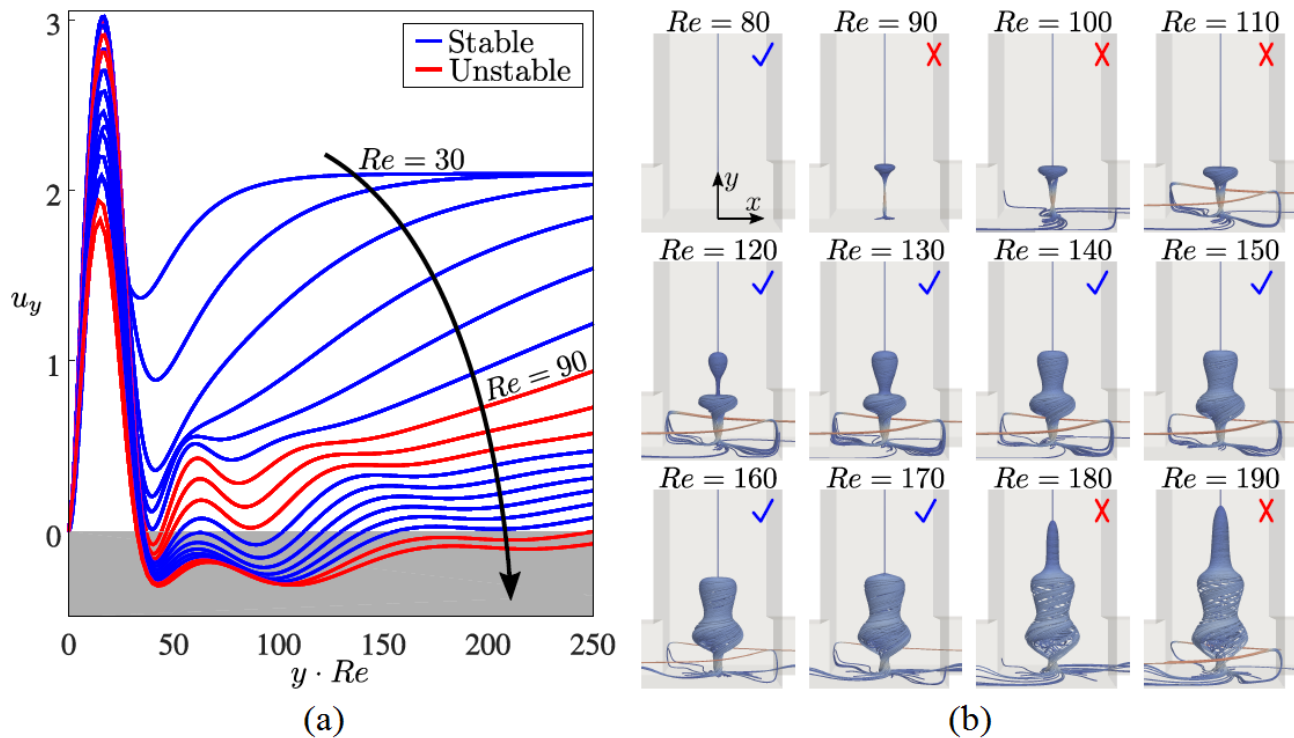


FIG. 6. Stability of the steady state solutions. (a) Distributions of dimensionless axial velocity  $u_y$  of the steady state solutions along the channel centerline. Increasing  $Re$  changes the curvature and level of  $u_y = u_y(yRe)$ , which in turn changes the number and positions of the internal stagnation points. (b) 3D streamline visualizations of the vortex breakdown regions in the outlet channel as a function of  $Re$ . The first bubble-type vortex breakdown region appears at around  $Re = 90$ . A second region then appears downstream of the first one at around  $Re = 120$ . These merge and become a single larger structure at around  $Re = 130$ . Finally, a third region appears further downstream at around  $Re = 180$  and enlarges and slenderizes the aforementioned second breakdown region. Blue check marks indicate stable solutions and red crosses indicate unstable solutions. Thus, the appearance of each subsequent vortex breakdown region results in a dynamical shift in the stability of the base state. That is, with zero vortex breakdown regions the flow is stable. When the first breakdown region appears, the flow becomes unstable. When the second breakdown region appears, the flow is restabilized. Finally, when the third breakdown region appears, the flow becomes unstable again. To the best of our knowledge, this is the first example of a flow in which the stability of the flow is so closely coupled to the appearance and dynamics of vortex breakdown.

second vortex breakdown region appears, which restabilizes the steady state flow solution. At around  $Re = 180$ , the third vortex breakdown region appears, triggering a helical instability in the vortex that causes a transition to the helically oscillating  $St \approx 1.75$  state. Thus, in the vortex T-mixer flow, stability characteristics of the flow are intimately coupled with the emergence and evolution of vortex breakdown. To the best of our knowledge, this is the first instance where this relationship has been identified to such a convincing extent.

Next, we seek to identify how and where the fluctuations grow on the steady state solutions. To achieve this, we probe the fluctuating velocity component  $|\delta u_z| = |u_z(\mathbf{x}, t) - u_z(\mathbf{x}, 0)|$  at  $\mathbf{x} = (0, 2, 0)$  as a function of time; for a linear instability, this perturbation velocity component should grow roughly exponentially over time. We also identify the location where the fluctuation originates by visualizing a level set of  $|\delta u| = |\mathbf{u}(\mathbf{x}, t) - \mathbf{u}(\mathbf{x}, 0)|$ . For the pulsating and helically oscillating states, we report results for the cases of  $Re = 100$  and  $180$ , respectively, which are unstable according to Fig. 6.

Results presenting the exponential growth of the fluctuating velocity components and the visualizations of the origins of the fluctuations are shown in Fig. 7. As can be seen, for both the pulsating and helically oscillating states, the per-

turbation velocity components grow exponentially in time, indicating that the instability is in fact a linear instability. For the  $Re = 100$  case, in which a single vortex breakdown bubble is present, the level set  $|\delta u| = 0.01$  appears within the junction of the vortex T-mixer, indicating that the fluctuation originates around the first vortex breakdown region. However, for the  $Re = 180$  case, in which three vortex breakdown regions are present, the level set  $|\delta u| = 0.01$  appears further downstream in the outlet channel, indicating that the fluctuation originates around the third vortex breakdown region. As this fluctuation grows, it extends upstream, suggesting that there is a pocket of absolute instability around the third vortex breakdown region. This renders the steady state solution globally unstable and triggers the helical  $St \approx 1.75$  oscillation that corresponds to spiral-type vortex breakdown [6]. Numerical results visualizing the growth of both oscillating states with particle tracers are presented in Appendix B.

As a final point of analysis, we apply the method of Ruith *et al.* [6] to calculate the criticality of the base state and demonstrate that, as expected, the appearance of vortex breakdown in the vortex T-mixer does correspond to the appearance of a pocket of subcriticality. This procedure involves axisymmetrizing the vortex data near the vortex core within a region where the flow is nominally axisymmet-

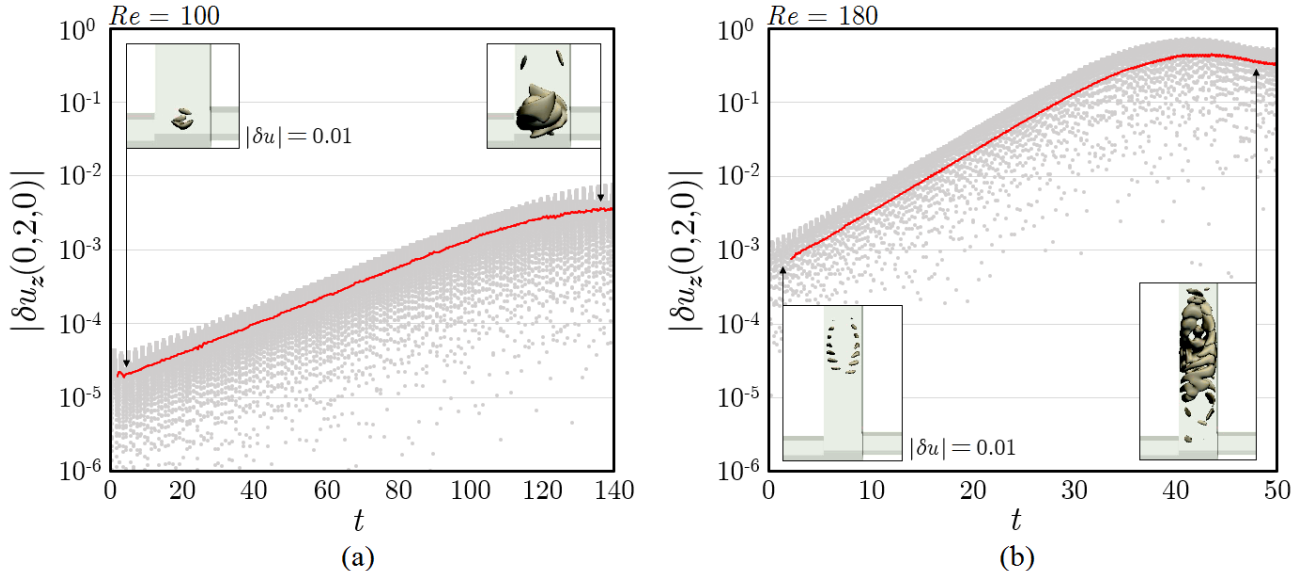


FIG. 7. Instability on the steady state solutions. (a) Growth of the  $St \approx 0.5$  fluctuation at  $Re = 100$ . The fluctuation  $|\delta u_z| = |u_z(\mathbf{x}, t) - u_z(\mathbf{x}, 0)|$  at  $\mathbf{x} = (0, 2, 0)$  grows exponentially with time  $t$ , indicating that the  $St \approx 0.5$  state is the result of a linear instability. A level set  $|\delta u| = |\mathbf{u}(\mathbf{x}, t) - \mathbf{u}(\mathbf{x}, 0)| = 0.01$  shows that the fluctuation originates around the first bubble breakdown region. (b) Growth of the  $St \approx 1.75$  fluctuation at  $Re = 180$ . The fluctuation  $|\delta u_z| = |u_z(\mathbf{x}, t) - u_z(\mathbf{x}, 0)|$  at  $\mathbf{x} = (0, 2, 0)$  grows exponentially with time  $t$ , indicating that the  $St \approx 1.75$  state is also the result of a linear instability. A level set  $|\delta u| = |\mathbf{u}(\mathbf{x}, t) - \mathbf{u}(\mathbf{x}, 0)| = 0.01$  shows that the fluctuation originates around the third bubble breakdown region. When  $|\delta u_z|$  reaches its plateau at sufficiently large  $t$ , the level set extends upstream, implying that there is a pocket of absolute instability around the third bubble breakdown region which generates the fluctuation.

ric. For simplicity, we select a cylindrical region about the vortex core with radius  $L/4$ . Next, the velocity is expressed in cylindrical coordinates, and we seek solutions to the differential equation given by [6]:

$$\left[ \frac{d^2}{dr^2} - \frac{1}{r} \frac{d}{dr} + \frac{1}{r^3 u_y^2} \frac{d(r u_\theta)^2}{dr} - \frac{r}{u_y} \frac{d}{dr} \left( \frac{1}{r} \frac{du_y}{dr} \right) \right] \phi_c = 0,$$

subject to the boundary conditions  $\phi_c(0) = 0$  and  $d\phi_c(0)/dr = 1$ . Critical values are sought such that  $\phi_c(r_{\text{crit}}) = 0$ , and the procedure is repeated for each axial location along the channel outlet. Finally, subcritical regions can be identified by plotting  $r_{\text{crit}}$  as a function of  $y$  and comparing it with a certain threshold. Numerical results for our data are presented in Fig. 8 for  $Re = 60, 90$ , and  $120$ , which correspond to Reynolds numbers for which we expect zero, one, and two vortex breakdown regions. As can be seen, for  $Re = 90$ ,  $r_{\text{crit}} \rightarrow 0$  at the same axial location where the first vortex breakdown region appears, and for  $Re = 120$ ,  $r_{\text{crit}} \rightarrow 0$  at approximately the same locations where the first two vortex breakdown regions appear. Despite the fact that several assumptions of the original theory presented by Ruith *et al.* [6] are less valid for our specific data due to the combination of flow geometry and Reynolds numbers, nevertheless, the application of this approach and the interpretation of vortex breakdown as due to the appearance of a pocket of subcriticality both appear to be well validated by this theory applied to the vortex T-mixer.

#### IV. CONCLUSIONS

The vortex T-mixer flow that we have studied here demonstrates several fluid dynamical features. We have identified a counter-intuitive flow regime in which increasing the Reynolds number restabilizes an unstable flow state. Critically, mechanism of this restabilization corresponds exactly to the appearance of a second vortex breakdown region that presumably restabilizes the flow through the nonlinear vortex breakdown dynamics. More generally, we have shown how the development and evolution of vortex breakdown in the vortex T-mixer are exactly coupled with the stability characteristics of the flow. The appearance of the first vortex breakdown region around  $Re = 90$  triggers a pulsative instability in the flow that launches the growth of a  $St \approx 0.5$  pulsating state. The appearance of a second vortex breakdown region around  $Re = 120$  restabilizes the base state. Finally, the appearance of a third vortex breakdown region around  $Re = 180$  triggers a helical instability in the flow that launches the growth of a helically oscillating  $St \approx 1.75$  state. Furthermore, the vortex T-mixer flow demonstrates flow regimes of multi-stability, including a regime of tri-stability for  $160 \leq Re \leq 170$  in which the steady, pulsating, and helically oscillating states are all stable. As future work, we plan to investigate in detail the specific mechanism by which the second vortex breakdown region stabilizes the flow using a global stability analysis. Furthermore, preliminary simulations with a cylindrical outlet suggest that the role of the corners may play a key role in this restabilization. Finally, a third oscillating state with  $St \approx 3.0$  was identified numerically, although its sensitivity to initial conditions prevented us from studying this state

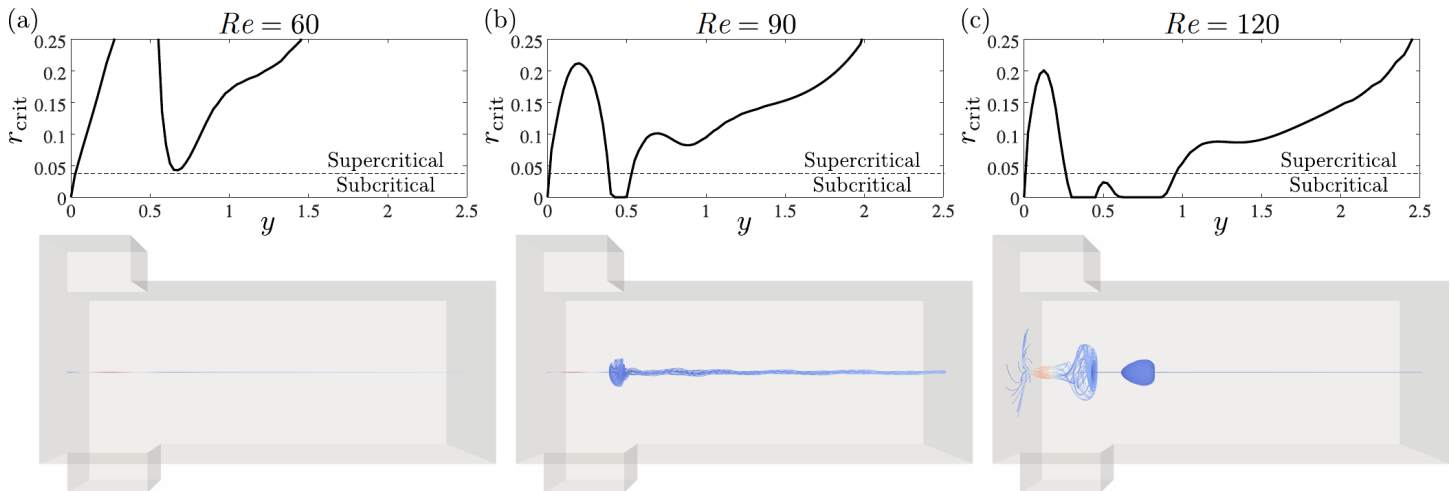


FIG. 8. Numerical results computing the criticality of our flows using the approach presented by Ruith *et al.* [6]. As can be seen, the appearance of a pocket of subcriticality corresponds to the appearance of vortex breakdown. Results correspond to (a)  $Re = 60$ , (b)  $Re = 90$ , and (c)  $Re = 120$ , for which we expect zero, one, and two regions of vortex breakdown in the flow. In each case, the axial locations where pockets of subcriticality appear correspond precisely with the locations of vortex breakdown.

experimentally. We hope to improve the sensitivity of our experimental approach in the future in order to verify and study this state. Ultimately, the vortex T-mixer flow that we have described here has demonstrated novel fluid dynamical features, the most important of which appears to be the close coupling between the vortex breakdown phenomenon and the stability characteristics of the flow, which to the best of our knowledge has not been documented elsewhere.

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### Appendix A: Simulation domain and convergence

Here, we briefly describe the simulation domain and mesh design used throughout our simulations as well as the boundary conditions used and the results from our numerical convergence test. Several representative views of the numerical simulation domain and mesh are presented in Fig. 9. All simulations were performed with approximately two million grid cells with inlets of length  $4L$  and an outlet of length  $15L$ , where the outlet channel cross section is  $L$  by  $L$ . At the inlets to our system, we specify the well-known fully developed velocity profile for a rectangular duct. With the coordinates defining the cross-section of a square channel given by  $y, z \in [-1, 1]$  and nondimensionalized by  $h_1$  and  $h_2$ , respectively, the fully developed axial velocity profile along a rectangular channel is given by [39,40]

$$u_x(y, z) = \frac{-1}{\frac{1}{3} - C^*} \left[ \frac{1}{2} (y^2 - 1) + \sum_{n=1}^{\infty} (-1)^{\frac{n-1}{2}} \frac{16}{n^3 \pi^3} \cos\left(\frac{n\pi}{2} y\right) \frac{\cosh\left(\frac{n\pi}{2} \frac{h_2}{h_1} z\right)}{\cosh\left(\frac{n\pi}{2} \frac{h_2}{h_1}\right)} \right], \quad \text{for } n \text{ odd} \quad (\text{A1})$$

where

$$C^* = \sum_{n=1}^{\infty} \frac{64}{n^5 \pi^5} \frac{h_1}{h_2} \tanh\left(\frac{n\pi}{2} \frac{h_2}{h_1}\right), \quad \text{for } n \text{ odd.} \quad (\text{A2})$$

Here, the axial velocity profile is nondimensionalized by the average axial flow speed. In order to set the Reynolds number, we scale the magnitude of this inlet velocity profile and adjust the kinematic viscosity in the solver.

A numerical convergence study was also performed to ensure that the flows were fully resolved. The results of this study are presented in Fig. 10. For the one million and two million grid cell cases, relative difference between the two peak velocities is less than 5%. We consider this as a nice enough convergence due to the large amount of transient

**simulations needed for the current study.** The convergence study was performed at the highest Reynolds number used in this study of  $Re = 250$ . At this Reynolds number, a third oscillating flow state appears with  $St \approx 3.0$ . However, we did not include this state in the main text because a sensitivity to initial conditions prevented us from identifying it experimentally. Nevertheless, since this corresponds to the highest Reynolds number from our study, we selected it for the numerical convergence study.

### Appendix B: Flow transitions

Here, we present numerical results demonstrating the transient fluid dynamics as the steady state flow transitions to the different oscillating states. Point particle tracers are generated near the origin  $\mathbf{x} = (0, 0, 0)$  and integrated forward in time as the flow evolves. Results demonstrating the transition from the steady state to the pulsating  $St \approx 0.5$  state are shown in Fig. 11, and results showing the transition from the steady state to the helically oscillating  $St \approx 1.75$  state are shown in Fig. 12. For the pulsating  $St \approx 0.5$  state, the oscillation begins to develop near the first vortex breakdown region in within the junction and propagates downstream as it grows, whereas for the helically oscillating  $St \approx 1.75$  state, the oscillation begins to develop downstream near the third vortex breakdown region and propagates upstream into the junction as it grows. For each case, the oscillation grows slowly at first so that even at  $t = 60$  the flows are still apparently steady visually. This is why time begins at  $t = 60$  in both figures.

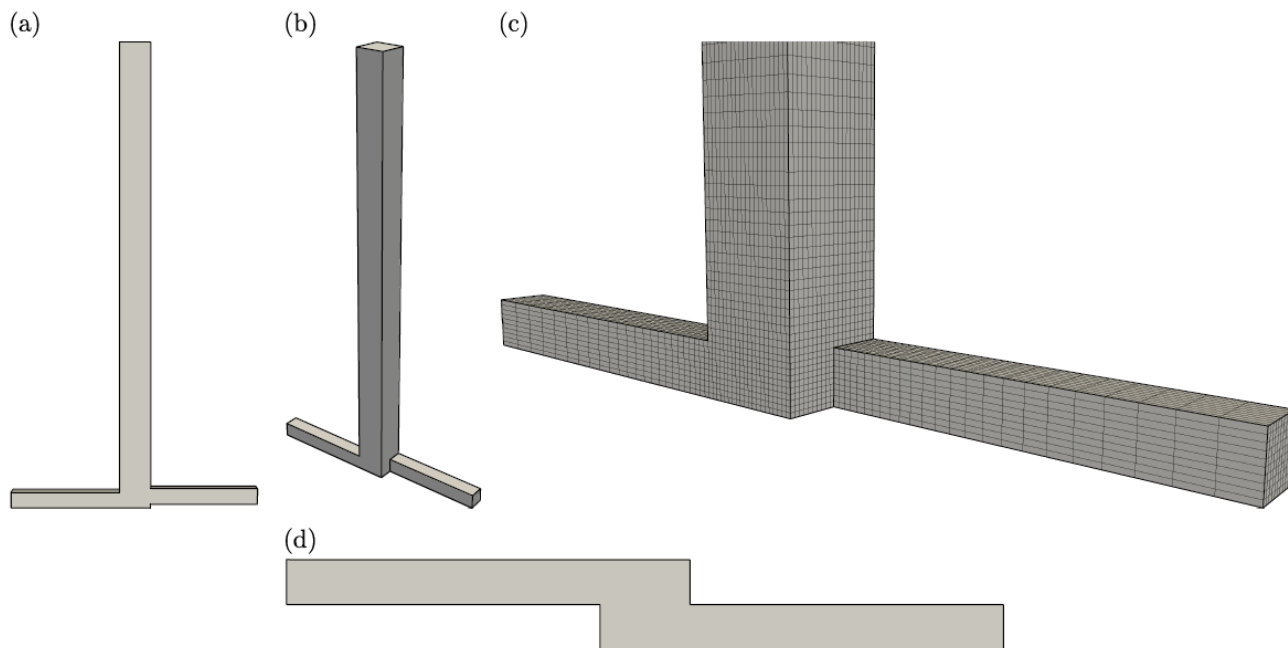


FIG. 9. Simulation domain visualized from (a) front view, (b) perspective view, (c) perspective view focused on the junction with outlined grid cells, and (d) bottom view. Grid cells have been purposefully coarsened by a factor of four in each direction for visualization purposes. High resolution simulations were performed with approximately two million grid cells with inlets of length  $4L$  and an outlet of length  $15L$ , where the outlet channel cross section is  $L$  by  $L$ . Cell lengths were graded towards the junction by factors of five and ten in the inlets and outlet, respectively.

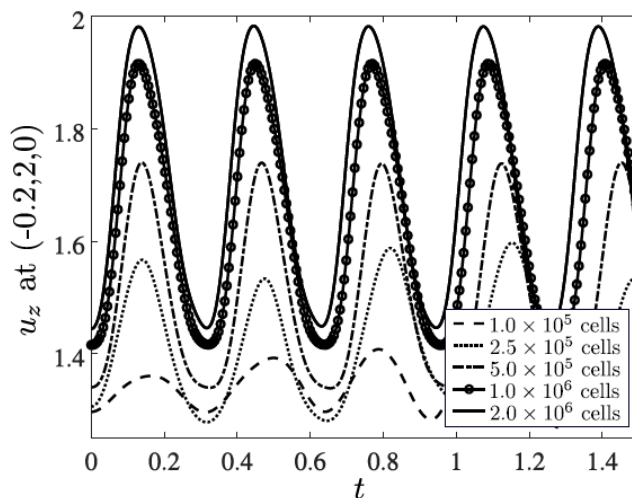


FIG. 10. Convergence test results show the oscillating velocity component  $u_z$  at  $(-0.2, 2, 0)$  for the  $St \approx 3.0$  state at  $Re = 250$ . The estimated Strouhal numbers are 3.47 with  $1.0 \times 10^5$  cells, 3.00 with  $2.5 \times 10^5$  cells, 3.04 with  $5.0 \times 10^5$  cells, 3.14 with  $1.0 \times 10^6$  cells, and 3.18 with  $2.0 \times 10^6$  cells. The relative error in the Strouhal number prediction with  $2.0 \times 10^6$  cells is less than 1.3%. Note that a third oscillating state with  $St \approx 3.0$  was identified numerically at higher Reynolds numbers, although it was not found experimentally due to sensitivity to the initial conditions. However, in order to ensure convergence of all the numerical results, we perform our convergence test on that state since it corresponded to the highest Reynolds number.

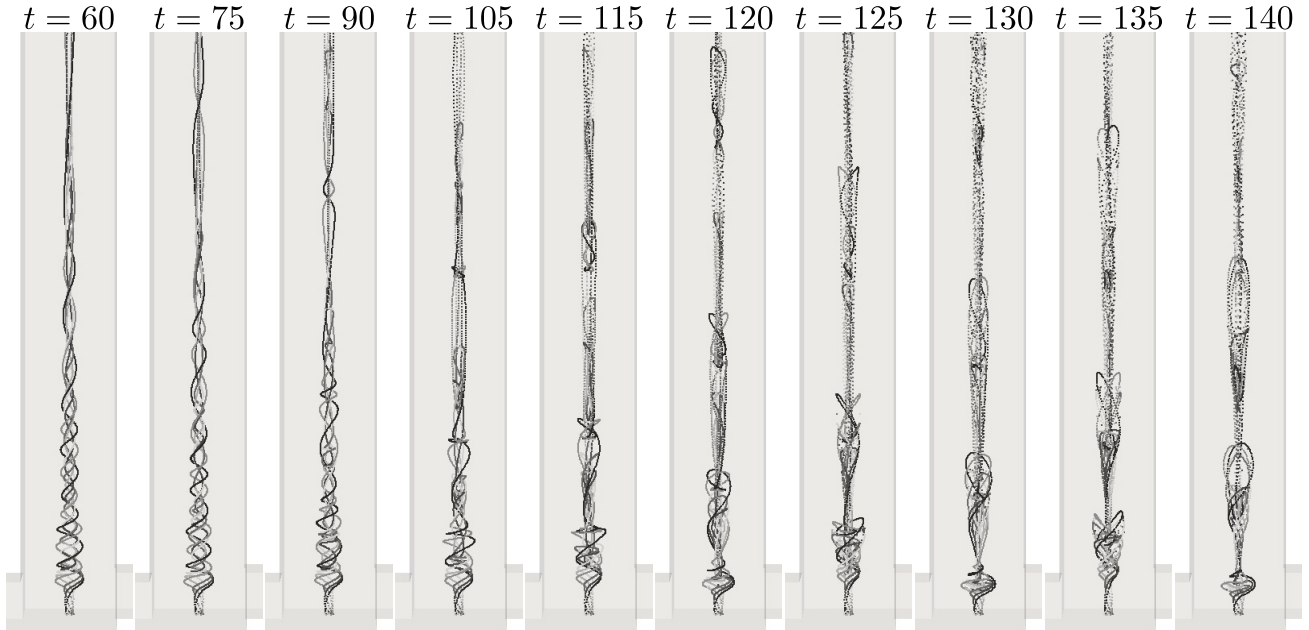


FIG. 11. Growth of the pulsative instability in the steady state solution at  $Re = 100$  as the flow transitions to the pulsating  $St \approx 0.5$  state. Fluid particles originating near  $\mathbf{x} = (0, 0, 0)$  are visualized over time  $t$ . Because the perturbation grows slowly, at  $t = 60$ , the flow is still visually steady. However, with increasing  $t$ , the pulsative instability develops upstream and propagates downstream in the outlet channel. At  $t = 140$ , the flow has approximately transitioned fully to the pulsatile  $St \approx 0.5$  state.

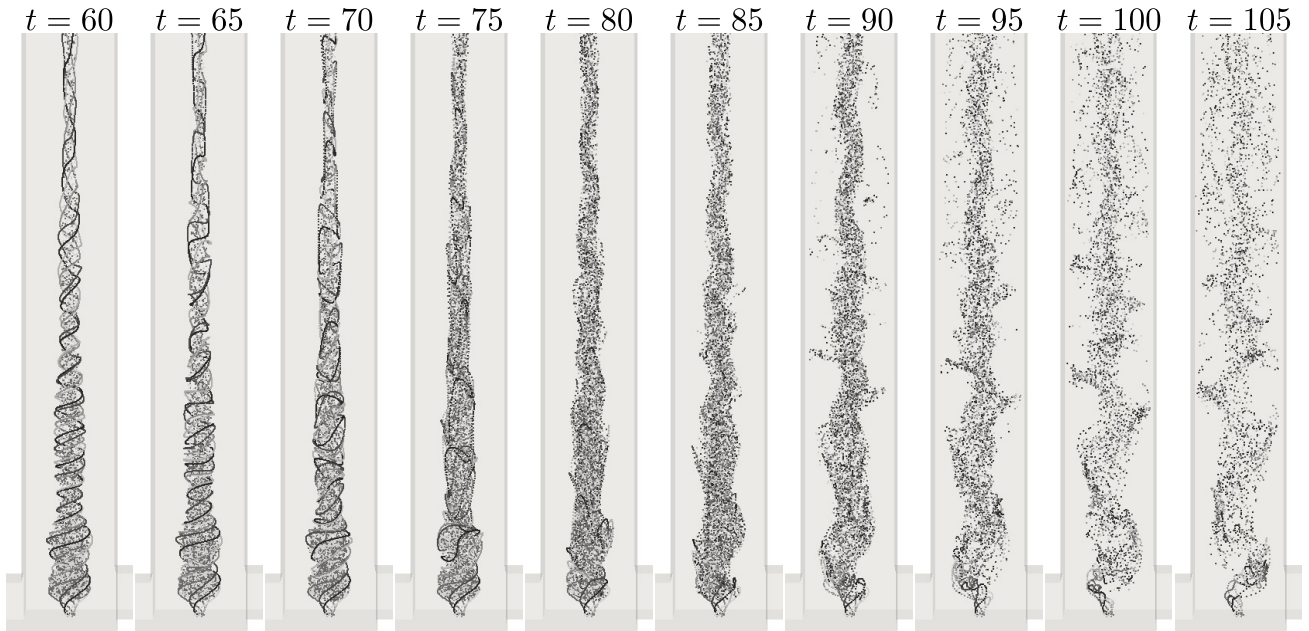


FIG. 12. Growth of the helical instability in the steady state solution at  $Re = 180$  as the flow transitions to the helically oscillating  $St \approx 1.75$  state. Fluid particles originating near  $\mathbf{x} = (0, 0, 0)$  are visualized over time  $t$ . Because the perturbation grows slowly, at  $t = 60$ , the flow is still visually steady. However, with increasing  $t$ , the helical instability develops downstream around the third vortex breakdown region and propagates upstream in the outlet channel. At  $t = 105$ , the flow has approximately transitioned fully to the helically oscillating  $St \approx 1.75$  state.

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